

REPORT DOCUMENTATION PAGE

Form Approved
OMB No. 0704-0188

Public reporting burden for this collection of information is estimated to average 1 hour per response, including the time for reviewing instructions, searching existing data sources, gathering and maintaining the data needed, and completing and reviewing this collection of information. Send comments regarding this burden estimate or any other aspect of this collection of information, including suggestions for reducing this burden to Department of Defense, Washington Headquarters Services, Directorate for Information Operations and Reports (0704-0188), 1215 Jefferson Davis Highway, Suite 1204, Arlington, VA 22202-4302. Respondents should be aware that notwithstanding any other provision of law, no person shall be subject to any penalty for failing to comply with a collection of information if it does not display a currently valid OMB control number. PLEASE DO NOT RETURN YOUR FORM TO THE ABOVE ADDRESS.

1. REPORT DATE (DD-MM-YYYY)		2. REPORT TYPE Technical Papers		3. DATES COVERED (From - To)	
4. TITLE AND SUBTITLE				5a. CONTRACT NUMBER	
				5b. GRANT NUMBER	
				5c. PROGRAM ELEMENT NUMBER	
6. AUTHOR(S)				5d. PROJECT NUMBER 2302	
				5e. TASK NUMBER MIG 2	
				5f. WORK UNIT NUMBER	
7. PERFORMING ORGANIZATION NAME(S) AND ADDRESS(ES) Air Force Research Laboratory (AFMC) AFRL/PRS 5 Pollux Drive Edwards AFB CA 93524-7048				8. PERFORMING ORGANIZATION REPORT	
9. SPONSORING / MONITORING AGENCY NAME(S) AND ADDRESS(ES) Air Force Research Laboratory (AFMC) AFRL/PRS 5 Pollux Drive Edwards AFB CA 93524-7048				10. SPONSOR/MONITOR'S ACRONYM(S)	
				11. SPONSOR/MONITOR'S NUMBER(S)	
12. DISTRIBUTION / AVAILABILITY STATEMENT Approved for public release; distribution unlimited.					
13. SUPPLEMENTARY NOTES					
14. ABSTRACT					
15. SUBJECT TERMS					
16. SECURITY CLASSIFICATION OF:			17. LIMITATION OF ABSTRACT A	18. NUMBER OF PAGES	19a. NAME OF RESPONSIBLE PERSON Leilani Richardson
a. REPORT Unclassified	b. ABSTRACT Unclassified	c. THIS PAGE Unclassified			19b. TELEPHONE NUMBER (include area code) (661) 275-5015

Standard Form 298 (Rev. 8-98)
Prescribed by ANSI Std. Z39.18

36 separate items are enclosed

G2

MEMORANDUM FOR PRR (Contractor/In-House Publication)

FROM: PROI (TI) (STINFO)

01 Aug 2000

SUBJECT: Authorization for Release of Technical Information, Control Number: **AFRL-PR-ED-TP-2000-158**
Wei, Y., Chow, C.L. (University of Michigan); Liu, C.T. (AFRL/PRSM), "Damage Analysis for Mixed Mode Crack Initiation"

International Conference on Computational Science
(Anaheim CA, 21-25 Aug 00)

(Statement A)
(Submission Deadline: 16 Aug 00)

1. This request has been reviewed by the Foreign Disclosure Office for: a.) appropriateness of distribution statement, b.) military/national critical technology, c.) export controls or distribution restrictions, d.) appropriateness for release to a foreign nation, and e.) technical sensitivity and/or economic sensitivity.

Comments: _____

Signature _____ Date _____

2. This request has been reviewed by the Public Affairs Office for: a.) appropriateness for public release and/or b) possible higher headquarters review.

Comments: _____

Signature _____ Date _____

3. This request has been reviewed by the STINFO for: a.) changes if approved as amended, b.) appropriateness of distribution statement, c.) military/national critical technology, d.) economic sensitivity, e.) parallel review completed if required, and f.) format and completion of meeting clearance form if required

Comments: _____

Signature _____ Date _____

4. This request has been reviewed by PR for: a.) technical accuracy, b.) appropriateness for audience, c.) appropriateness of distribution statement, d.) technical sensitivity and economic sensitivity, e.) military/national critical technology, and f.) data rights and patentability

Comments: _____

APPROVED/APPROVED AS AMENDED/DISAPPROVED

LAWRENCE P. QUINN
Technical Advisor
Rocket Propulsion Division

DATE

20021119 134

Damage Analysis for Mixed-Mode Crack Initiation

? hyphenated in conclusions...

Y. Wei and C. L. Chow

Department of Mechanical Engineering
University of Michigan-Dearborn
Dearborn, MI 48128

C. T. Liu
AFRL/PRSM
10 E. Saturn Blvd.
Edwards AFB CA 93524-7680

Distribution A: Approved for Public Release

~~OBJECTIVE~~

Introduction ?

Prediction of mixed mode fracture load and crack initiation angle for Al 2024-T3 and particulate composite material with the theory of damage mechanics

Objectives

- 1) Develop a model to characterize damage in a material element
- 2) Propose a failure criterion with the concept of damage accumulation
- 3) Implement the damage model into ABAQUA through UMAT subroutine
- 4) Apply the model for mixed mode fracture analysis

Font size
got bigger
here

DAMAGE MODEL

Relationship Between Effective and True Stress

$$\bar{\sigma} = \mathbf{M}(\mathbf{D}) : \sigma$$

The damage effect tensor $\mathbf{M}(\mathbf{D})$ is expressed with two damage variables D and μ as

$$\mathbf{M} = \frac{1}{1-D} \begin{bmatrix} 1 & \mu & \mu & 0 & 0 & 0 \\ \mu & 1 & \mu & 0 & 0 & 0 \\ \mu & \mu & 1 & 0 & 0 & 0 \\ 0 & 0 & 0 & 1-\mu & 0 & 0 \\ 0 & 0 & 0 & 0 & 1-\mu & 0 \\ 0 & 0 & 0 & 0 & 0 & 1-\mu \end{bmatrix}$$

DAMAGE COUPLED CONSTITUTIVE EQUATION

The elastic law of damaged material

$$\varepsilon^e = \mathbf{C}^{-1} : \sigma \qquad \mathbf{C}^{-1} = \mathbf{M} : \mathbf{C}_0^{-1} : \mathbf{M}$$

\mathbf{C}_0 and \mathbf{C} are elastic tensors respectively for undamaged and damaged materials

The yield surface is postulated with damage consideration as

$$F_p(\sigma, R) = \frac{1-\mu}{1-D} \sigma_{eq} - [R_0 + R(p)] = 0$$

σ_{eq} the Von-Mises equivalent stress

The plastic law of damaged material

$$d\varepsilon^p = \lambda_p \frac{\partial F_p}{\partial \sigma} \qquad dp = \lambda_p \frac{\partial F_p}{\partial (-R)} = \lambda_p$$

DAMAGE SURFACE

The plastic damage surface is

$$F_d(Y, B) = Y_d - [B_0 + B(w)] J = 0 \quad Y_d = \left\{ \frac{1}{2} (Y_D^2 + \gamma Y_\mu^2) \right\}^{\frac{1}{2}}$$

Y_D, Y_μ the thermodynamic conjugate forces of the damage variables D and μ

B_0 the initial damage threshold

B the damage hardening

w the equivalent damage

γ the damage-related material constant

DAMAGE EVOLUTION

$$dD = -\lambda_d \frac{\partial F_d}{\partial Y_D} = -\frac{\lambda_d Y_D}{2Y_d}$$

$$d\mu = -\lambda_d \frac{\partial F_d}{\partial Y_\mu} = -\frac{\lambda_d \gamma Y_\mu}{2Y_d}$$

$$dw = -\lambda_d \frac{\partial F_d}{\partial B} = \lambda_d$$

λ_d the Lagrange multiplier

FINITE ELEMENT FORMULATION

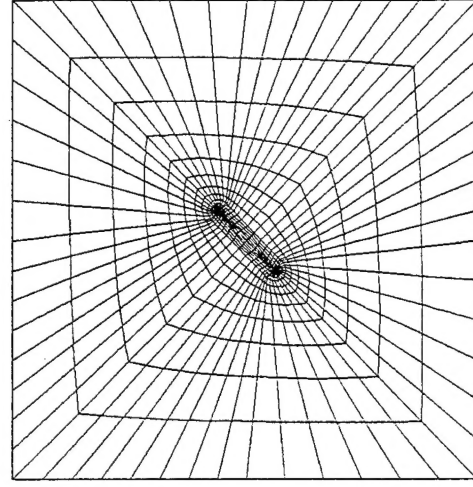
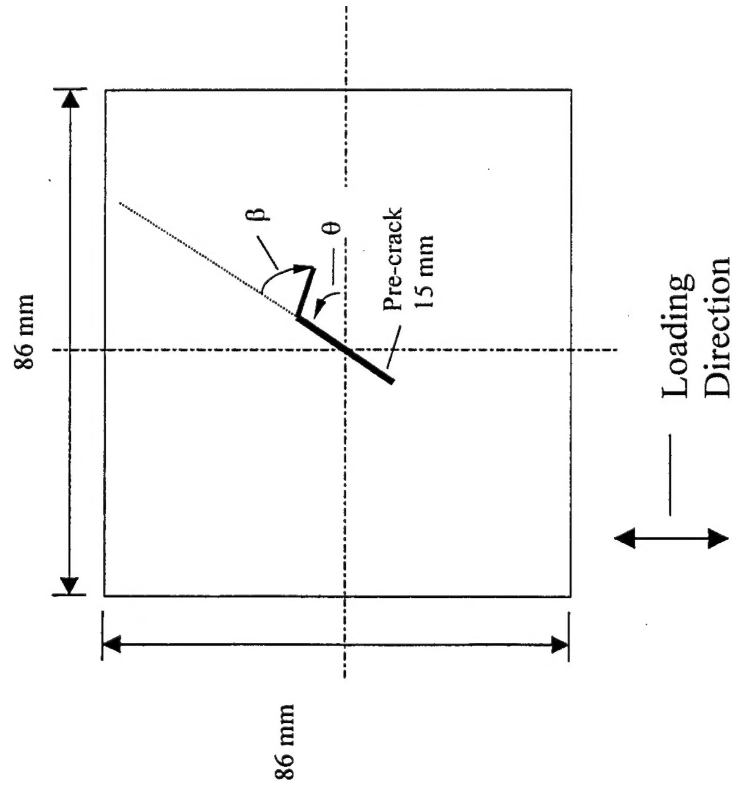
The proposed damage model is discretized and coded in the user subroutine UMAT of a finite element package ABAQUS (version 5.8).

$$d\sigma = \mathbf{C}^{ep} : d\boldsymbol{\varepsilon} \qquad \mathbf{C}^{ep} = \mathbf{M}^{T,-1} : \mathbf{U}^{T,-1} : \mathbf{C}_0^{ep} : \mathbf{M}^{T,-1}$$

\mathbf{C}^{ep} the effective instantaneous tangent modulus tensor

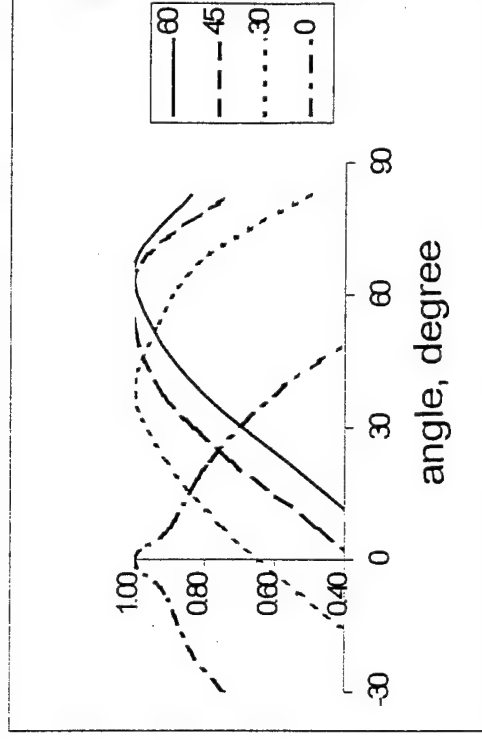
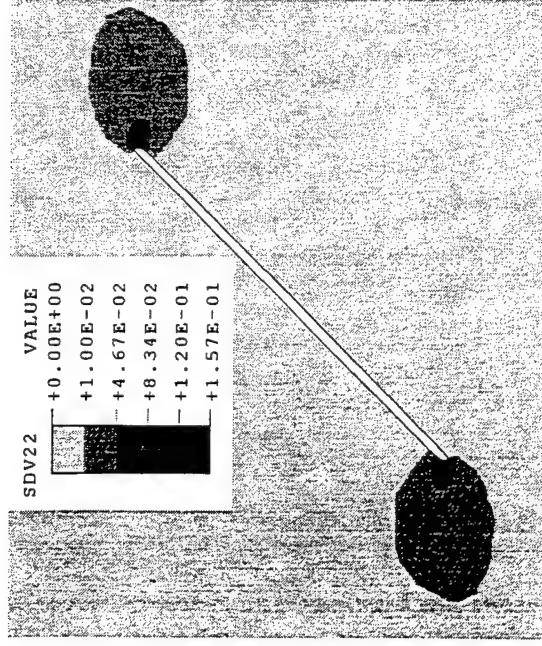
$$\begin{aligned} \mathbf{U} &= \mathbf{I} - \left(\mathbf{U}_0 + \mathbf{C}_0^{ep} : \mathbf{M}^{T,-1} : \mathbf{U}_0 : \mathbf{M}^{T,-1} : \mathbf{C}^{-1} \right) : \boldsymbol{\sigma} : \mathbf{T} : \mathbf{M}^{T,-1} \\ \mathbf{U}_0 &= \frac{\partial \mathbf{M}}{\partial D} \frac{\partial F_d}{\partial Y_D} + \frac{\partial \mathbf{M}}{\partial \mu} \frac{\partial F_d}{\partial Y_\mu} \end{aligned}$$

Crack Initiation Angles for Al 2024-T3 Plates



Typical finite elements
for mixed-mode fracture analysis
?

Crack Initiation Angles for Al 2024-T3 Plates



Damage distribution contours in
AL2024-T3 plate for $\theta = 45^\circ$

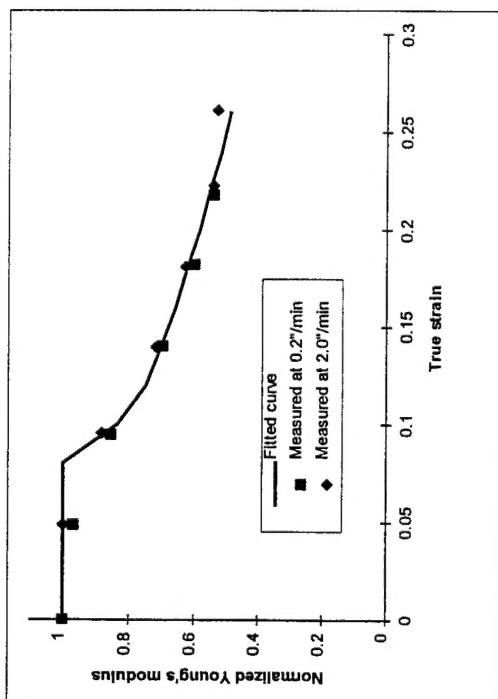
Angular distributions of damage for
mixed-mode AL2024-T3 specimen

Crack Initiation Angles for Al 2024-T3 Plates

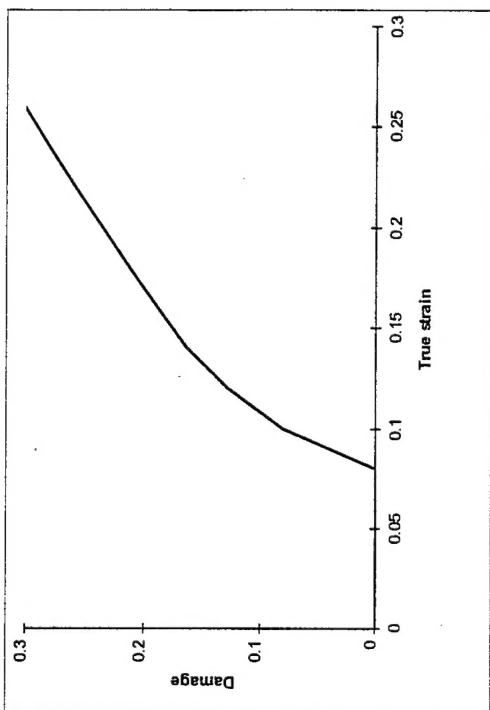
Table 1 Crack Initiation Angle β_i for Al2024-T3

inclined angle θ	test	numerical simulation	
		proposed model	Chow&Wang's model
0	0	0	0
30	35.9	37.5	43
45	53.7	52.5	56
60	71.2	67.5	73

Measurement for Particulate Composite Plates

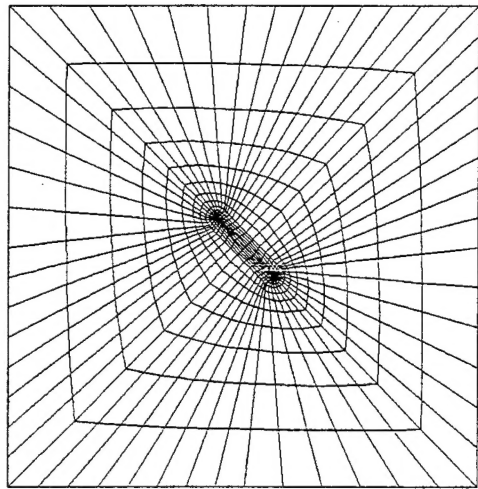
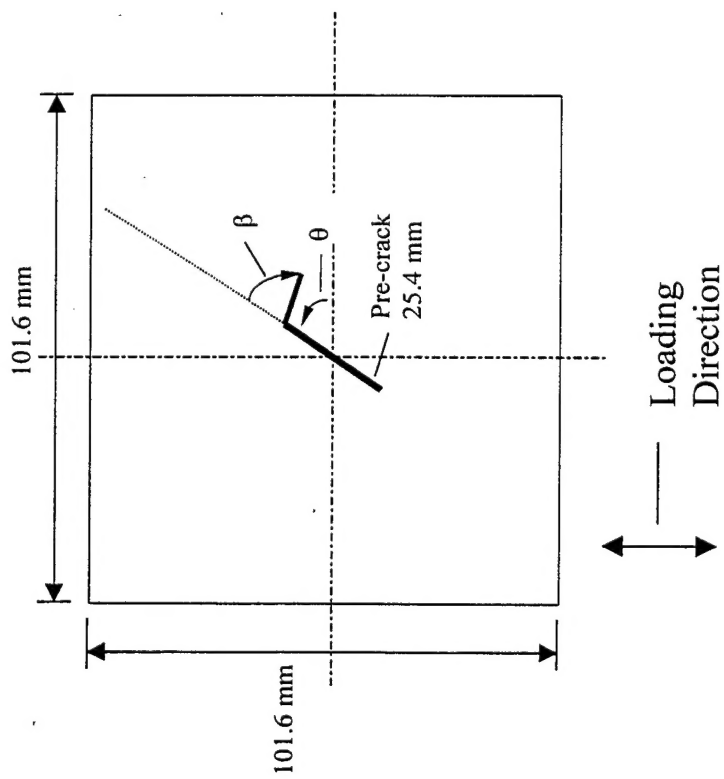


Normalized Young's modulus vs. applied strain for a particulate composite



Damage evolution curve for a particulate composite

Fracture Analysis for Particulate Composite Plates



Typical finite elements
for mixed mode fracture analysis

?

Fracture Analysis for Particulate Composite Plates

Table 2 Crack Initiation Load for Particulate Composite

Pre-crack angle (°)		0	30	60
Load (lb)	prediction	24.0	27.2	36.9
	test	23.4	27.0	36.2

Table 3 Crack initiation Angle β_i (°) Particulate Composite

Pre-crack angle θ (°)		0	30	60
β_i (°)	prediction	0	28	62
	test	0	33	68

Conclusions

- (1) Numerical modeling result indicates that damage accumulation is confined around the crack tip region.
- (2) For 2024-T3 Al, both the isotropic and the anisotropic damage models can be used to predict the crack initiation loads and angles under mixed-mode loadings with good accuracy.
- (3) For the particulate composite material, the predicted crack initiation loads and angles, based on the proposed isotropic damage model, agree well with the test results.
- (4) Both the crack initiation loads and angles increase with increasing the initial crack inclined angles.